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# Joint Altitude, Beamwidth, Location and Bandwidth Optimization for UAV-Enabled Communications

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*Abstract*—This letter investigates an uplink power control problem for unmanned aerial vehicles (UAVs) assisted wireless communications. We jointly optimize the UAV's flying altitude, antenna beamwidth, UAV's location and ground terminals' allocated bandwidth and transmit power to minimize the sum uplink power subject to the minimal rate demand. An iterative algorithm is proposed with low complexity to obtain a suboptimal solution. Numerical results show that the proposed algorithm can achieve good performance in terms of uplink sum power saving.

*Index Terms*—UAV communications, altitude optimization, beamwidth optimization, location placement, bandwidth allocation.

#### I. INTRODUCTION

Unmanned aerial vehicles (UAVs) assisted wireless communications have attracted considerable attention recently due to its maneuverability and increasing affordability [1]. Compared to conventional wireless communications, UAV-enabled wireless communications can provide higher wireless connectivity in areas without infrastructure coverage and achieve higher capacity for line-of-sight (LoS) communication links with the ground terminals (GTs).

To fully exploit the design degrees of freedom for UAVenabled communications, it is crucial to investigate the UAV mobility in the three-dimensional space. In [2], the altitude of UAV was optimized to provide maximum radio coverage on the ground. For an underlaid device-to-device communication network with one UAV, the optimal values for the UAV altitude were analyzed in [3] for the maximum system sum rate and coverage probability. Considering the adjustable UAVs' locations over time, the UAV number and trajectory optimization problems were respectively considered in [4] and [5]. Further optimizing user-UAV association, [6] investigated the sum power minimization problem of the UAV. Different from [2]-[6] with fixed-beamwidth antenna, the beamwidth of the directional antenna and the altitude of the UAV were jointly optimized in [7] to improve the system throughput. However, the optimal beamwidth was only examined numerically and simple equal bandwidth allocation was assumed in [7], even

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though proper bandwidth allocation can further enhance the system performance.

In this letter, we aim to minimize the sum power for an uplink UAV-enabled wireless communication. There are two main contributions. One contribution is that we consider joint altitude, beamwidth, location and bandwidth allocation, and an algorithm is proposed by solving three subproblems iteratively, where each subproblem can be solved optimally. The other contribution is to effectively obtain the optimal beamwidth with the bisection method when the pathloss exponent is two, and to obtain the optimal solution in closed form for bandwidth allocation subproblem.

#### **II. SYSTEM MODEL AND PROBLEM FORMULATION**

Consider an uplink UAV-enabled wireless communication system with one flying UAV and K GTs. The UAV is deployed as a flying BS with horizontal and vertical location  $\boldsymbol{y} = (y(1), y(2))$  at hight H. The horizontal and vertical location of GT k is denoted by  $\boldsymbol{x}_k = (x_k(1), x_k(2))$ , and the hight of each GT is assumed to be zero compared with the hight of the UAV.

Assume that the UAV is equipped with a directional antenna with adjustable beamwidth, while each GT is equipped with an omnidirectional antenna with unit gain. The azimuth and elevation half-power beamwidths of the UAV antenna are equal, which are both denoted by  $2\Theta \in (0, \pi)$ . According to [8, Eq. (2-51)], the antenna gain in the direction with azimuth angle  $\theta$  and elevation angle  $\phi$  can be modeled as

$$G = \begin{cases} \frac{G_0}{\Theta^2} & \text{if } 0 \le \theta \le \Theta \text{ and } 0 \le \phi \le \Theta \\ g \approx 0 & \text{otherwise,} \end{cases}$$
(1)

where  $G_0 \approx 2.2846$ , and g means the channel gain outside the beamwidth of the antenna. For simplify, we set g = 0. We consider the case that the GTs are located outdoors, and the channel between each GT and the UAV is mainly a LoS path. The uplink channel gain between GT k and the UAV is

$$g_k = \frac{g_0}{(\|\boldsymbol{y} - \boldsymbol{x}_k\|^2 + H^2)^{\frac{\alpha}{2}}},$$
 (2)

where  $\|\cdot\|$  denotes the Euclidian norm,  $g_0$  is the channel power gain at the reference distance 1 m, H is the hight of the UAV,  $\left(\|\boldsymbol{y} - \boldsymbol{x}_k\|^2 + H^2\right)^{\frac{1}{2}}$  is the distance between GT k and the UAV, and  $\alpha \geq 2$  is the pathloss exponent. Based on (1) and (2), the uplink achievable rate of GT k in the coverage area of the UAV is

$$r_{k} = w_{k} \log_{2} \left( 1 + \frac{p_{k}g_{0}G_{0}}{w_{k}\sigma^{2}\Theta^{2}(\|\boldsymbol{y} - \boldsymbol{x}_{k}\|^{2} + H^{2})^{\frac{\alpha}{2}}} \right), \quad (3)$$

where  $w_k$  is the allocated bandwidth for GT k,  $p_k$  is the transmit power of GT k,  $\sigma^2$  is the noise power density and

 $w_k \sigma^2$  is the noise power for decoding the information of GT k at the UAV side. For GT k, the minimal rate constraint  $r_k \ge R_k$  should be satisfied. Since we aim at minimizing uplink sum power of all GTs, it is always energy saving to transmit with minimal rate. Setting  $r_k = R_k$  in (3), we have

$$p_{k} = aw_{k} \left(2^{\frac{R_{k}}{w_{k}}} - 1\right) \Theta^{2} \left(\|\boldsymbol{y} - \boldsymbol{x}_{k}\|^{2} + H^{2}\right)^{\frac{\alpha}{2}}, \quad (4)$$

where  $a = \frac{\sigma^2}{g_0 G_0}$ . We aim at minimizing the uplink sum power of all GTs whilst satisfying the minimal rate constraints. Mathematically, the sum power minimization problem is

$$\min_{\boldsymbol{H},\boldsymbol{\Theta},\boldsymbol{y},\boldsymbol{w}} \sum_{k=1}^{K} a w_k \left( 2^{\frac{R_k}{w_k}} - 1 \right) \boldsymbol{\Theta}^2 \left( \|\boldsymbol{y} - \boldsymbol{x}_k\|^2 + H^2 \right)^{\frac{\alpha}{2}}$$
(5a)

s.t. 
$$aw_k \left(2^{\frac{R_k}{w_k}} - 1\right) \Theta^2 \left(\|\boldsymbol{y} - \boldsymbol{x}_k\|^2 + H^2\right)^{\frac{\alpha}{2}} \le P_k,$$
  
 $\forall k = 1, \cdots, K$  (5b)

$$\|\boldsymbol{y} - \boldsymbol{x}_k\|^2 \le H^2 \tan^2 \Theta, \quad \forall k = 1, \cdots, K$$
 (5c)

$$\sum_{k=1}^{K} w_k \le B \tag{5d}$$

$$H_{\min} \le H \le H_{\max}, \Theta_{\min} \le \Theta \le \Theta_{\max}$$
 (5e)

$$w_k \ge 0, \quad \forall k = 1, \cdots, K.$$
 (5f)

where  $\boldsymbol{w} = (w_1, \dots, w_K)$ , *B* is the maximal bandwidth of the system,  $P_k$  is the maximum transmit power of GT *k*,  $[H_{\min}, H_{\max}]$  is the feasible region of height *H* determined by obstacle heights and authority regulations, and  $[\Theta_{\min}, \Theta_{\max}]$ is the feasible region of half-beamwidth determined by practical antenna beamwidth tuning technique. Constraints in (5c) ensure that all GTs are in the coverage area of the UAV.

### III. PROPOSED ALGORITHM

Due to nonconvex objective function (5a) and constraints (5b)-(5c), Problem (5) is a nonconvex problem. It is generally hard to obtain the globally optimal solution to Problem (5). To solve this problem, we propose an iterative algorithm with low complexity through sequently optimizing  $(H, \Theta)$ ,  $\boldsymbol{y}$  and  $\boldsymbol{w}$ . It is fortunate that we can globally optimize each variable with other variables fixed in each step.

#### A. Optimal Altitude and Beamwidth

With fixed y and w, Problem (5) is formulated as

$$\min_{H,\Theta} \qquad \sum_{k=1}^{K} A_k \Theta^2 (D_k + H^2)^{\frac{\alpha}{2}} \tag{6a}$$

s.t. 
$$A_k \Theta^2 (D_k + H^2)^{\frac{\alpha}{2}} \le P_k, \quad \forall k = 1, \cdots, K$$
 (6b)  
 $H^2 \tan^2 \Theta \ge D$  (6c)

II 
$$\tan \Theta \ge D_{\max}$$
 (00)

$$H_{\min} \le H \le H_{\max}, \Theta_{\min} \le \Theta \le \Theta_{\max},$$
 (6d)

where  $A_k = aw_k \left(2^{\frac{R_k}{w_k}} - 1\right)$ ,  $D_k = ||\boldsymbol{y} - \boldsymbol{x}_k||^2$ , and  $D_{\max} = \max_{k=1,\dots,K} D_k$ . Denoting  $H^*$  as optimal value of H and observing that the objective function (6a) is an increasing function in H with given  $\Theta$ , we can claim that

$$H^* = \max\left\{\frac{\sqrt{D_{\max}}}{\tan\Theta}, H_{\min}\right\},\tag{7}$$

for the optimal solution. This claim can be proved by the contradiction method. If  $(H, \Theta)$  is the optimal solution of Problem (6) with  $H > H^*$ , we find that solution  $(H^*, \Theta)$  is also a feasible solution of Problem (6) with  $\sum_{k=1}^{K} A_k \Theta^2 (D_k + (H^*)^2)^{\frac{\alpha}{2}} < \sum_{k=1}^{K} A_k \Theta^2 (D_k + H^2)^{\frac{\alpha}{2}}$ , which contradicts the hypothesis that  $(H, \Theta)$  is the optimal solution. Based on (7), we consider the value of  $H^*$  in the following two cases.

1) Case 1: With  $H^* = H_{\min}$ , Problem (6) is equivalent to

$$\min_{\Theta} \quad \Theta \tag{8a}$$

s.t. 
$$A_k \Theta^2 (D_k + H_{\min}^2)^{\frac{\alpha}{2}} \le P_k, \quad \forall k = 1, \cdots, K$$
 (8b)  
 $H_{\min}^2 \tan^2 \Theta > D_{\max}$  (8c)

$$\Theta_{\min} \le \Theta \le \Theta_{\max}.$$
(8d)

Since Problem (8) is a minimization of  $\Theta$ , the optimal solution is thus

$$\Theta^* = \max\left\{\arctan\frac{\sqrt{D_{\max}}}{H_{\min}}, \Theta_{\min}\right\},\tag{9}$$

which is the minimal value of  $\Theta$  satisfying (8b), (8c) and (8d). Note that Problem (8) is feasible if and only if

$$\Theta^* \le \min\left\{\min_{k=1,\cdots,K} \sqrt{\frac{P_k}{A_k (D_k + H_{\min}^2)^{\frac{\alpha}{2}}}}, \Theta_{\max}\right\}.$$
 (10)

2) Case 2: With  $H^* = \frac{\sqrt{D_{\text{max}}}}{\tan \Theta}$ , Problem (6) becomes

$$\min_{\Theta} \qquad \sum_{k=1}^{K} A_k \Theta^2 \left( D_k + \frac{D_{\max}}{\tan^2 \Theta} \right)^{\frac{1}{2}} \tag{11a}$$

s.t. 
$$A_k \Theta^2 \left( D_k + \frac{D_{\max}}{\tan^2 \Theta} \right)^{\frac{1}{2}} \le P_k, \quad \forall k = 1, \cdots, K$$
(11b)

$$H_{\min}^2 \tan^2 \Theta \le D_{\max} \tag{11c}$$

$$\Theta_{\min} \le \Theta \le \Theta_{\max}. \tag{11d}$$

Due to the complicated objective function (11a), it is generally difficult to obtain the optimal  $\Theta^*$  of Problem (11) in closed form.  $\Theta^*$  can be obtained via a one-dimension exhaustive search over  $[\Theta_{\min}, \Theta_{\max}]$ .

Specifically, for the special case where pathloss exponent  $\alpha = 2$ , we can fortunately obtain the optimal  $\Theta^*$  through a simple bisection method. When the GTs are located outdoors in rural areas, and the communication channel between the UAV and each GT is dominated by the LoS path, i.e.,  $\alpha = 2$  [7, Eq. (2)]. For  $\alpha = 2$ , we define function

$$f_k(x) = x^2 \left( D_k + \frac{D_{\max}}{\tan^2 x} \right), \quad x \in [0, \pi/2),$$
 (12)

$$h_1(x) = (\cot x - x - x \cot^2 x) \cot x,$$
 (13)

$$h_2(x) = x + 2x\cos^2(x) - 3/2\sin(2x),$$
 (14)

and then we have

$$f'_{k}(x) = 2x \left( D_{k} + D_{\max} h_{1}(x) \right), \qquad (15)$$

$$h_1'(x) = \csc^4(x)h_2(x),$$
 (16)

$$h'_2(x) = -4(x\cos(x) - \sin(x))\sin(x),$$
 (17)

- If  $D_k \frac{2}{3}D_{\max} \ge 0$ , then  $D_k + D_{\max}h_1(x) \ge 0$  for all  $x \in [0, \pi/2)$ , i.e.,  $f_k(x)$  is monotonically increasing.
- If  $D_k \frac{2}{3}D_{\max} < 0$ , there must exist one solution  $x_k$  such that  $D_k + D_{\max}h_1(x_k) = 0$  due to the fact that  $D_k \ge$ 0 and  $h_1(x)$  is a continuous function. In this situation,  $f_k(x)$  first decreases for  $x \in [0, x_k]$  and then increases with  $x \in (x_k, \pi/2)$ .

According to the above analysis,  $f_k(\Theta) \leq P_k$  is equivalent to  $\Theta_k^{\min} \leq \Theta \leq \Theta_k^{\max}$ , where  $\Theta_k^{\min}$  and  $\Theta_k^{\max}$  can be obtained by using the bisection method. As a result, constraints (11b)-(11d) can be equivalently transformed to

$$\bar{\Theta}_{\min} \le \Theta \le \bar{\Theta}_{\max},$$
 (18)

where  $\bar{\Theta}_{\min} = \max\{\max_{k=1,\dots,K} \Theta_k^{\min}, \Theta_{\min}\}, \bar{\Theta}_{\max} =$  $\min\{\min_{k=1,\dots,K} \Theta_k^{\max}, \Theta_{\max}, \arctan(\sqrt{D_{\max}}/H_{\min})\}.$ 

Based on the definition of  $f_k(x)$ , the objective function (11a) can be expressed as  $f(\Theta) = \sum_{k=1}^{K} f_k(\Theta)$ . We have  $f'(\Theta) = \sum_{k=1}^{K} f'_k(\Theta) = 2x(\sum_{k=1}^{K} D_k + KD_{\max}h_1(\Theta))$ . To show the monotonicity of  $\Theta$  in  $f_{\Theta}$ show the monotonicity of  $\Theta$  in  $[\overline{\Theta}_{\min}, \overline{\Theta}_{\max}]$ , we also consider the following three situations:

- If  $\sum_{k=1}^{K} D_k + KD_{\max}h_1(\bar{\Theta}_{\min}) \ge 0$ , then  $\sum_{k=1}^{K} D_k + KD_{\max}h_1(\Theta) \ge 0$  for  $\Theta \in [\bar{\Theta}_{\min}, \bar{\Theta}_{\max}]$ , i.e.,  $f(\Theta)$ is monotonically increasing. The optimal beamwidth is  $\Theta^* = \bar{\Theta}_{\min}.$
- Θ<sup>\*</sup> = Θ<sub>min</sub>.
  If Σ<sup>K</sup><sub>k=1</sub> D<sub>k</sub> + KD<sub>max</sub>h<sub>1</sub>(Θ
  max) < 0, f(Θ) is monotonically decreasing and Θ<sup>\*</sup> = Θ
  max.
  If Σ<sup>K</sup><sub>k=1</sub> D<sub>k</sub> + KD<sub>max</sub>h<sub>1</sub>(Θ
  max) < 0 and Σ<sup>K</sup><sub>k=1</sub> D<sub>k</sub> + KD<sub>max</sub>h<sub>1</sub>(Θ
  max) > 0, there must exist one solution Θ
  such that Σ<sup>K</sup><sub>k=1</sub> D<sub>k</sub> + KD<sub>max</sub>h<sub>1</sub>(Θ) = 0. In this situation f(Θ) first decreases for x ∈ [Θ
  mix σ] and then situation,  $f(\overline{\Theta})$  first decreases for  $x \in [\overline{\Theta}_{\min}, \overline{\Theta}]$  and then increases with  $x \in (\overline{\Theta}, \overline{\Theta}_{\max}]$ , i.e.,  $\Theta^* = \overline{\Theta}$ .

Note that Problem (11) is feasible if and only if  $\overline{\Theta}_{\min} \leq$  $\bar{\Theta}_{max}$ . By comparing the objective values of the solutions obtained in the above two cases, the one with lower objective value is chosen as the optimal solution to Problem (6).

#### B. Optimal Location Planning

For Problem (5) with fixed  $(H, \Theta)$  and  $\boldsymbol{w}$ , the location planning problem can be formulated as

$$\min_{\boldsymbol{y}} \sum_{k=1}^{K} C_k (\|\boldsymbol{y} - \boldsymbol{x}_k\|^2 + H^2)^{\frac{\alpha}{2}}$$
(19a)

s.t. 
$$\|\boldsymbol{y} - \boldsymbol{x}_k\|^2 + H^2 \le \bar{E}_k, \quad \forall k = 1, \cdots, K$$
 (19b)

$$\|\boldsymbol{y} - \boldsymbol{x}_k\|^2 \le H^2 \tan^2 \Theta, \quad \forall k = 1, \cdots, K,$$
 (19c)

where  $C_k = aw_k \left(2^{\frac{R_k}{w_k}} - 1\right)\Theta^2$ , and  $\bar{E}_k = \left(\frac{P_k}{C_k}\right)^{\frac{2}{\alpha}}$ . Since  $\|\boldsymbol{y} - \boldsymbol{x}_k\|^2$  is a convex function and  $x^{\frac{\alpha}{2}}$  is convex and nondecreasing,  $(\|\boldsymbol{y} - \boldsymbol{x}_k\|^2 + H^2)^{\frac{\alpha}{2}}$  is convex based on the scalar composition property of convex functions [9]. As a result, Problem (19) is a convex problem, which can be effectively solved via the standard interior point method.

## C. Optimal Bandwidth Allocation

It remains to investigate the bandwidth allocation with fixed location, altitude and beamwidth. Define function  $u_k(x) =$  $x2^{\frac{R_k}{x}} - x$  for x > 0, and we have

$$u_k'(x) = 2^{\frac{R_k}{x}} - \frac{(\ln 2)R_k 2^{\frac{R_k}{x}}}{x} - 1, u_k''(x) = \frac{(\ln 2)^2 R_k^2 2^{\frac{R_k}{x}}}{x^3} > 0$$
(20)

From (20), we observe that function  $u_k(x)$  is a convex function, which indicates that Problem (5) is a convex problem with fixed  $(H, \Theta)$  and **y**. Based on (20), we have  $u'_k(x) <$  $\lim_{x\to+\infty} u'_k(x) = 0$ , i.e.,  $u_k(x)$  is a monotonically decreasing function, which is helpful in transforming constraints (5b). With optimized  $(H, \Theta)$  and  $\boldsymbol{y}$ , Problem (5) is equivalent to

$$\min_{\boldsymbol{w}} \qquad \sum_{k=1}^{K} F_k w_k \left( 2^{\frac{R_k}{w_k}} - 1 \right) \tag{21a}$$

$$\text{.t.}\qquad \sum_{k=1}^{K} w_k \le B \tag{21b}$$

$$k \geq W_k \quad \forall k = 1, \cdots, K,$$
 (21c)

where  $F_k = a\Theta^2 (\|\boldsymbol{y} - \boldsymbol{x}_k\|^2 + H^2)^{\frac{\alpha}{2}}, W_k = u_k^{-1} (\frac{P_k}{b_k})$ , and  $u_k^{-1}(x)$  is the inverse function of  $u_k(x)$ . The lagrangian of convex Problem (21) is

$$\mathcal{L}(\boldsymbol{w},\lambda) = \sum_{k=1}^{K} F_k w_k \left( 2^{\frac{R_k}{w_k}} - 1 \right) + \lambda \left( \sum_{k=1}^{K} w_k - B \right), \quad (22)$$

where  $\lambda$  is the non-negative Lagrange multiplier associated with constraint (21b). According to [9] and [10, Appendix A], the KKT conditions of (21) are

$$\frac{\partial \mathcal{L}}{\partial w_k} = F_k \left( 2^{\frac{R_k}{w_k}} - \frac{(\ln 2)R_k}{w_k} 2^{\frac{R_k}{w_k}} - 1 \right) + \lambda = 0$$
(23)

From (23), we have

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$$\lambda = F_k \left( -e^{\frac{(\ln 2)R_k}{w_k}} + \frac{(\ln 2)R_k}{w_k} e^{\frac{(\ln 2)R_k}{w_k}} + 1 \right).$$
(24)

Define function  $u(x) = xe^{x} - e^{x} + 1$ , x > 0. We have u'(x) = $xe^x > 0, \forall x > 0$ . Thus, function u(x) is strictly increasing and u(x) > u(0) = 0,  $\forall x > 0$ . Based on (24) and (21c), we have

$$w_k = \max\left\{\frac{(\ln 2)R_k}{u^{-1}\left(\frac{\lambda}{F_k}\right)}, W_k\right\}, \quad \forall k = 1, \cdots, K, \quad (25)$$

where  $u^{-1}(x)$  is the inverse function of u(x). According to (24),  $\lambda = F_k u \left( \frac{(\ln 2)R_k}{w_k} \right) > 0$ , which implies that (21b) holds with equality. Plugging (25) into (21b) yields

$$B = \sum_{k=1}^{K} \max\left\{\frac{(\ln 2)R_k}{u^{-1}\left(\frac{\lambda}{F_k}\right)}, W_k\right\} \triangleq \hat{u}(\lambda).$$
(26)

Equation (26) has a unique solution  $\lambda > 0$ . Since u(x)is strictly increasing, inverse function  $u^{-1}(x)$  is also strictly increasing in  $(0, +\infty)$ . Thus,  $\hat{u}(\lambda_i)$  is a strictly decreasing function in  $(0, +\infty)$ . Owing to the fact that  $\lim_{\lambda\to 0+} \hat{u}(\lambda) =$  $+\infty$  and  $\lim_{\lambda\to +\infty} \hat{u}(\lambda) = 0$ , there exists one unique  $\lambda$ satisfying  $\hat{u}(\lambda) = B$ , and the solution can be obtained by using the bisection method. Having obtained the value of  $\lambda$ , the optimal  $\boldsymbol{w}$  can be obtained from (25).

D. Iterative Algorithm and Complexity Analysis

Algorithm 1: Iterative Algorithm

1: Set the initial solution  $(H^{(0)}, \Theta^{(0)}, \boldsymbol{y}^{(0)}, \boldsymbol{w}^{(0)})$ , and iteration number n = 1.

2: repeat

- 3: With fixed  $\boldsymbol{y}^{(n-1)}$  and  $\boldsymbol{w}^{(n-1)}$ , obtain the optimal  $(H^{(n)}, \Theta^{(n)})$  of problem (6).
- 4: With fixed  $(H^{(n)}, \Theta^{(n)})$  and  $\boldsymbol{w}^{(n-1)}$ , obtain the optimal  $\boldsymbol{y}^{(n)}$  of problem (19).
- 5: With fixed  $(H^{(n)}, \Theta^{(n)})$  and  $\boldsymbol{y}^{(n)}$ , obtain the optimal  $\boldsymbol{w}^{(n)}$  of problem (21).
- 6: Set n = n + 1.
- 7: **until** the objective function (5a) converges.

The iterative procedure for solving Problem (5) is given in Algorithm 1. The main complexity of Algorithm 1 lies in Problem (6) and Problem (21). For Problem (6), the major computation comes from case 2, which needs to solve Problem (11) via a one-dimension exhaustive search method with complexity  $\mathcal{O}(\frac{\Theta_{\max}-\Theta_{\min}}{\delta})$  for minimal step  $\delta$ . To solve Problem (21), the major complexity lies in solving (26) with complexity  $\mathcal{O}(\log_2(1/\epsilon))$  for the bisection method with accuracy  $\epsilon$ . As a result, the total complexity of Algorithm 1 is  $\mathcal{O}(L_{it}\frac{\Theta_{\max}-\Theta_{\min}}{\delta}+L_{it}\log_2(1/\epsilon))$ , where  $L_{it}$  is the number of iterations of the iterative Algorithm 1.

#### **IV. NUMERICAL RESULTS**

We consider that there are K = 20 GTs uniformly distributed in a circular area with radius 300 m. We set  $g_0 = 1.42 \times 10^{-4}$ , B = 10 MHz,  $P_1 = \cdots = P_K = 20$ dBm,  $\sigma^2 = -169$  dBm/Hz,  $H_{\min} = 50$  m,  $H_{\max} = 500$  m,  $\Theta_{\min} = 0$ , and  $\Theta_{\max} = \pi/2$  rad. We consider equal minimal rate demand, i.e.,  $R_1 = \cdots = R_K = R$ .

In Fig. 1, we consider the sum power (11a), which equals to (5a) with fixed y and w, versus  $\Theta$  for various minimal rate demands with equal bandwidth allocation and the UAV located at the center of the circle. With given minimal rate demand, it is observed that the sum power first decreases and then increases with the increase of  $\Theta$ , which verifies the theoretical analysis in Section III.A.

We compare the proposed algorithm with the following four methods: fixed location method with optimized altitude, beamwidth and bandwidth (labeled as 'FL'), fixed altitude and beamwidth method with optimized location and bandwidth (labeled as 'FAB'), fixed bandwidth method with optimized location altitude and beamwidth (labeled as 'FB'), and exhaustive method via running Algorithm 1 with 1000 initial points (labeled as 'Exhaustive'). We investigate the sum power versus the minimal rate demand in Fig. 2. It can be seen that the proposed algorithm outperforms FL, FAB and FB, especially when the minimal rate demand is large. This is because the proposed algorithm jointly optimizes altitude,



Fig. 1. Sum power versus  $\Theta$ .



Fig. 2. Sum rate versus the minimal rate demand with  $\alpha = 2$ .

beamwidth, location and bandwidth. It can be seen that the sum power of the exhaustive method is slightly lower than that of the proposed algorithm, which indicates that the proposed algorithm approaches the globally optimal solution.

#### V. CONCLUSION

In this letter, we investigated the sum power minimization problem in uplink UAV-enabled communications. We showed that the sum power first decreases and then decreases with the beamwidth. Numerical results showed that the uplink sum power performance can be improved by the proposed algorithm.

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